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Dispersionless Manipulation of Reflected Acoustic Wavefront by Subwavelength Corrugated Surface

Yi-Fan Zhu^{1,2,*}, Xin-Ye Zou^{1,2,*}, Rui-Qi Li¹, Xue Jiang¹, Juan Tu¹, Bin Liang^{1,2} & Jian-Chun Cheng^{1,2,3}

Free controls of optic/acoustic waves for bending, focusing or steering the energy of wavefronts are highly desirable in many practical scenarios. However, the dispersive nature of the existing metamaterials/metasurfaces for wavefront manipulation necessarily results in limited bandwidth. Here, we propose the concept of dispersionless wavefront manipulation and report a theoretical, numerical and experimental work on the design of a reflective surface capable of controlling the acoustic wavefront arbitrarily without bandwidth limitation. Analytical analysis predicts the possibility to completely eliminate the frequency dependence with a specific gradient surface which can be implemented by designing a subwavelength corrugated surface. Experimental and numerical results, well consistent with the theoretical predictions, have validated the proposed scheme by demonstrating a distinct phenomenon of extraordinary acoustic reflection within an ultra-broad band. For acquiring a deeper insight into the underlying physics, a simple physical model is developed which helps to interpret this extraordinary phenomenon and predict the upper cutoff frequency precisely. Generations of planar focusing and non-diffractive beam have also been exemplified. With the dispersionless wave-steering capability and deep discrete resolution, our designed structure may open new avenue to fully steer classical waves and offer design possibilities for broadband optical/acoustical devices.

The recent decade has witnessed a rapid development in the field of wavefront manipulation research¹⁻¹⁹. In conventional optical and acoustic components, the wavefront shaping is achieved on the basis of reflection, refraction or diffraction of waves. The emergence of metasurfaces provides a new way to manipulate wavefront freely by introducing abrupt changes of optical/acoustical phases, which breaks the dependence on the propagation effect, and enables molding wavefront into arbitrary shapes with sub-wavelength resolution². Optical metasurfaces have exhibited great potentials to achieve novel phenomena such as extraordinary refraction/reflection³⁻⁶, coupling of propagating wave into evanescent waves⁷, planar focusing⁸⁻¹⁰, non-diffracting beams¹¹, and vortex¹². Acoustic counterparts have also been realized with similar capability of acoustic wavefront manipulation¹³⁻¹⁹. Despite the tremendous prospect of these devices with extraordinary wave manipulation flexibilities, they are inevitably frequency dependent and have limited bandwidth, in which the desired phase response of a subunit is designed only for a specific frequency and, moreover, the transverse size of a supercell needs to be related to wavelength¹⁻¹⁹. As a

²Key Laboratory of Modern Acoustics, MOE, Institute of Acoustics, Department of Physics, Nanjing University, Nanjing 210093, P. R. China. ²Collaborative Innovation Center of Advanced Microstructures, Nanjing University, 210093, P. R. China. ³State Key Laboratory of Acoustics, Chinese Academy of Sciences, Beijing 100190, P. R. China. ^{*}These authors contributed equally to this work. Correspondence and requests for materials should be addressed to B.L. (email: liangbin@nju.edu.cn) or J.C.C. (email: jccheng@nju.edu.cn)

result, the change of frequency will unavoidably lead to different phase profiles. So far, the manipulation of optic/acoustic waves with unlimited bandwidth is still challenging and remains an open question.

Here, we propose the concept of dispersionless wavefront manipulation and develop a general scheme to design reflective surface with a hitherto inaccessible functionality of going beyond the restriction of bandwidth. A specific implementation by subwavelength corrugated surface (SCS) has been demonstrated theoretically and experimentally to mimic the desired gradient profile, both showing the distinct property of extraordinary reflection in an ultra-broad band. A simple model based on the phased array theory is presented to interpret this extraordinary phenomenon. By applying the same design method, more complicated phase profiles can be realized via the gradient surface with subwavelength resolution. We anticipate our designed structure with distinct mechanism will constitute a significant step towards practicable wavefront-controlling devices with broadband functionalities and have enlightening significance in the field of optical and acoustical wavefront engineering.

Results

Concept of dispersionless phase front. For the purpose of realizing dispersionless wavefront manipulation, we start from the fundamental difficulty of eliminating the frequency dependence in generalized law of reflection, which is deduced by Fermat's principle and generally governs the behavior of the reflected wave at surface with a spatially-varying phase abrupt change introduced, as follows²

$$\sin \theta_r - \sin \theta_i = \frac{\lambda}{2\pi} \frac{d\phi}{dx},\tag{1}$$

where θ_r and θ_i are the angles of reflection and incidence, respectively. λ is wavelength and $\phi = \phi(x)$ is the phase response at the surface. The reflected angle θ_r can be calculated as

$$\theta_r = \arcsin\left[\sin(\theta_i) + \frac{\lambda}{2\pi} \frac{d\phi}{dx}\right] \tag{2}$$

It is expected from Eq. (2) that various phase fronts can be yielded by designing the function of $d\phi/dx$ appropriately. For instance, one can realize extraordinary reflection by choosing $d\phi/dx$ as a constant everywhere on the interface, while planar focusing requires a specific $d\phi/dx$ that varies with x to control the convergence of wave. Actually, these are the common ways the wavefront manipulations are achieved in previous designs. However, the optic/acoustic elements employed for realizing the desired phase gradient, are usually frequency selective, leading to the wavelength-dependence of the term $d\phi/dx$. Then the term $(\lambda/2\pi)(d\phi/dx)$ seems to be necessarily a function of the incident wavelength, which means θ_r must differ at different frequencies¹⁻¹⁹. As a result, the conventional wavefront-manipulation devices always have dispersion characteristics and limited bandwidth. Although some efforts have been dedicated to the design of broadband devices, their bandwidth is broadened by simply assuming an $d\phi/dx$ invariant with respect to frequency, making it impossible to realize a constant reflected angle for real broadband signals^{6,9}.

Observation of Eq. (2) suggests, however, that it is quite possible to exploit a frequency-dependent but appropriately-designed phase profile, i.e., $d\phi/dx$, to compensate the dispersion in the factor $(\lambda/2\pi)$ perfectly, rather than leading to a highly dispersive wavefront manipulation as expected conventionally. For making the term $(\lambda/2\pi)(d\phi/dx)$ independent of frequency, the phase gradient $d\phi/dx$ should have the following form

$$\frac{d\phi}{dx} = \frac{2\pi a(x)}{\lambda},\tag{3}$$

where a(x) is a function of x and independent of λ .

The above analyses reveal the potential to realize dispersionless wavefront manipulation via design of a specific gradient surface. Figure 1(a) gives the schematic illustration of dispersionless phase front, which shows that the desired phase gradient $d\phi/dx$ should be proportional to the wave number $k_0 = 2\pi/\lambda$, i.e., $\phi(x) = k_0 \int a(x) dx$. Apparently, the presence of the factor of k_0 is for vanishing the frequency dependence, while the remaining term (a(x)) that depends on x exclusively, can still be tailored freely to yield arbitrary wavefront, as will be discussed later.

In the following, we show how the SCS allows arbitrary control over the reflected phase front with the specific condition given by Eq. (3) being satisfied. Figure 1(b) shows the schematic of SCS. By decorating the surface with grooves that has subwavelength width, the phase of the wave is delayed when propagating along a round trip in each groove. Compared with a normal flat surface, the decorated one adds a local phase shift to the reflected wave. The relationship between input and output phases is simply determined by the wave path, as follows

$$\phi(x) = \phi_{out} - \phi_{in} = -\frac{4\pi h(x)}{\lambda}, \tag{4}$$

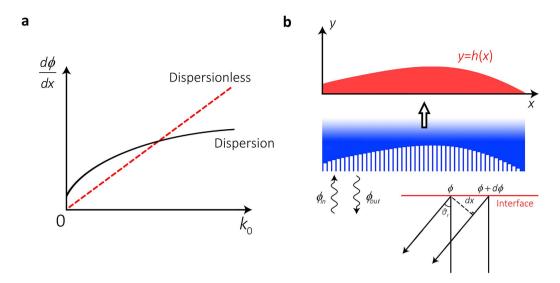


Figure 1. Schematic illustration of dispersionless SCS. (a) Schematic illustration of dispersionless phase front. A dispersionless phase front should have $d\phi/dx$ proportional to k_0 . (b) Schematic illustration of SCS made of grooves with subwavelength width below the surface. Blue region is filled with acoustically rigid medium and the background medium is chosen as air. The depth of grooves is a function of x, i.e., y = h(x), which determines the spatially-varying phase shift between incidence and reflection.

where h(x) is the depth of groove, and $\phi(x)$ represents the spatial phase response at the surface. In the proposed SCS structure, the dimension of the grooves along x-direction is subwavelength, which guarantees a subwavelength resolution of phase profile along the surface. On the other hand, the depth of the grooves along y-direction is comparable to the working wavelengths in order to achieve enough phase retardation. Thus, by choosing the depths of the grooves at each spatial position differently, spatially-varying phase shift can be obtained for the reflected wave. It is noteworthy that the phase response given by Eq. (4) agrees with Eq. (3) in form, indicating that the requirement for eliminating the frequency-related term is met excellently. The reflected wave direction can be deduced by Eq. (2)

$$\theta_r = \arcsin[\sin(\theta_i) + 2g(x)], \tag{5}$$

where g(x) = dh(x)/dx is the spatial gradient of the groove array. In such cases, the reflected direction is independence of wavelength and solely dominated by the choice of g(x). Thus, by designing g(x) appropriately, arbitrary phase front can be achieved in broad band, as will be demonstrated later.

Ultra-broadband extraordinary reflection by SCS. The following analyses take a particular extraordinary reflection SCS (ERSCS) with constant gradient of groove arrays (viz., g(x)) as example. Figure 2(a) shows the structural parameters of the ERSCS with g=0.3535. For an incident wave with frequency of $f_0=c_0/2h_{18}=2695\,\mathrm{Hz}$ ($c_0=343\,m/s$ is sound speed in air), the range of the phase shift provided by the grooves is $0-2\pi$ with step of $\pi/9$. Therefore, for an incident wave with arbitrary frequency of $\beta f_0.(\beta>0)$, the reflected phase can be also modulated linearly between $0-2\beta\pi$ with step of $\beta\pi/9$, which means the requirement for yielding the extraordinary reflection can always be fulfilled at different frequencies.

Figures 2(b,c) show the photograph of the sample and the 2-D schematic of the experimental system, respectively. The measurements were performed in the anechoic chamber. To obtain a plane wave, a loudspeaker was located 3 m away from the sample. The measuring area and the center of loudspeaker are located in the same *x-y* plane. The reflected angle deduced from Eq. (5) for normally-incident wave is 45°. An absorptive plate was placed in between to separate the incident and reflected acoustic fields. The error caused by diffraction effect near the edge of the plate should be negligible since both the sizes of ERSCS and the measuring area are much larger than the incident wavelength.

We have performed a series of measurements within a broadband, and the typical results will be presented as follows. Numerical simulations have also been carried out for comparison. The numerical and the experimental results are illustrated in Fig. 3 for three particular frequencies: $2.7f_0$, $4.5f_0$, and $6.3f_0$ (viz., $7.277\,\text{kHz}$, $12.128\,\text{kHz}$ and $16.979\,\text{kHz}$). A parameter of phase number (PN), which represents the number of discrete phases within 2π range, is defined as PN = n/β to evaluate the resolution of the phase surface. For the three considered frequencies, the PN values are 6.67, 4, and 2.86, respectively. Good agreement is observed between the numerical and experimental results, in which the extraordinary reflection of 45° , exactly the same as the theoretical prediction, can be identified within

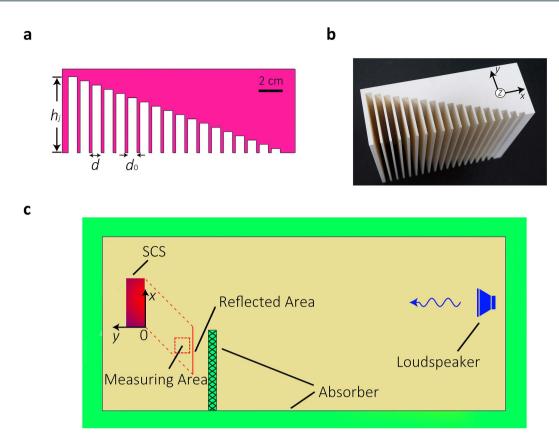


Figure 2. Schematic illustration of ERSCS and experimental setup. (a) Schematic cross-section of ERSCS made of 18 grooves, with period $d=1\mathrm{cm}$ and width $d_0=0.75\mathrm{cm}$. The depth h_j of each groove increases from 3.535 mm to 63.63 mm with step of 3.535 mm. The size in x-y plane is 19.5 cm \times 7 cm. (b) A photograph of the ERSCS sample made of acrylonitrile butadiene styrene (ABS) plastic manufactured by 3D printer. The size is $19.5\,\mathrm{cm}\times7\,\mathrm{cm}\times18\,\mathrm{cm}$. (c) Schematic illustration of 2-D experimental system. The measuring area is a 6 cm \times 6 cm square region centered at (-8 cm, -16 cm). The 'Reflected Area' is defined as a 19.5 cm wide cross profile with its center coordinate ($-9.75\,\mathrm{cm}$, $-19.5\,\mathrm{cm}$). An absorptive plate was placed between source and measuring region to separate the incident and reflected acoustic fields.

an ultra-broadband frequency range. It is worth mentioning that reflected angle yielded by the proposed structure can be controlled freely within a remarkably wide range (approximately $-80^{\circ}-+80^{\circ}$). Note also that the effect of extraordinary reflection can be maintained in the cases of oblique incidence because the grooves with subwavelength width only permit zeroth-order mode to propagate, making the phase response independent of the incident direction. We have also performed a series of numerical simulations to estimate the effectiveness of the proposed structure for oblique incident waves and the numerical results have verified that this extraordinary phenomenon also occurs with high efficiency in oblique incidence cases. (See Supplementary Materials).

Physical model based on phased array theory. The simplicity in the structure of this particular ERSCS enables an analytical analysis for better understanding of its underlying mechanism, which can be accomplished by exploiting the phased array theory. At orifice of an individual groove, the incident wave will induce a vibration of air that behaves as a secondary radiation source for which the strength factor A_j is estimated approximately by $A_j = Sv_j$, where v_j is normal velocity at the orifice and S is cross-section of each groove. Due to the subwavelength width of groove, we can simply consider the abnormal reflected wave as a new acoustic radiation by a line of secondary sources with phase delays modulated by the SCS with varying groove depths, as shown in Fig. 4(a). Assuming the distance between observation point and the jth source is ρ_j , the total reflected pressure can be expressed as

$$p(\rho, \theta, \omega) = \sum_{j=1}^{N} A_j \exp(i\phi_j) H_0^{(1)}(k_0 \rho_j),$$
 (6)

where $H_0^{(1)}$ is zeroth order Hankel function of the first kind and ϕ_j is the phase delay of each secondary source. When the incident wave is a plane wave and width of each groove is same, all of the strength

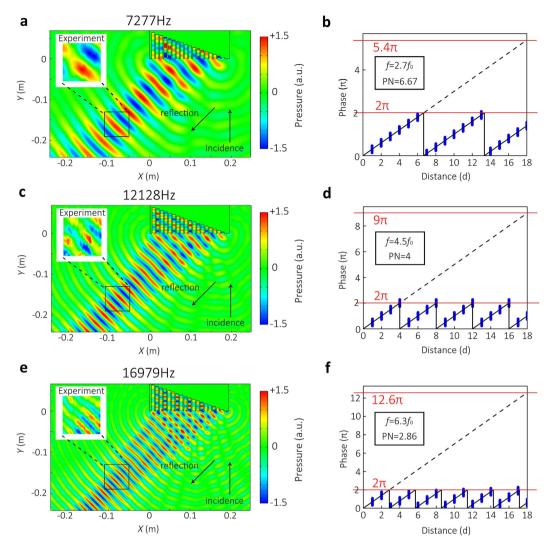


Figure 3. Numerical and experimental demonstrations of ultra-broadband extraordinary reflection by SCS. (a,c,e) Scattered acoustic pressure fields of ERSCS in simulation at 7.277 kHz, 12.128 kHz, and 16.979 kHz, respectively. The insets show the sound fields measured in the in the measuring area of Fig. 2(c). (b, d, f) The corresponding phase distributions along the interface at 7.277 kHz, 12.128 kHz, and 16.979 kHz, respectively. The phase keeps a linear dependence on the distance for different frequencies.

factors should be the same, i.e., $A_j \equiv A$. Since the phase delay by each groove is $\phi_j = 2k_0h_j$ with $h_j = (N - i)dg$, (N = n + 1 = 19), at far-field Eq. (6) becomes

$$p(\rho, \theta, \omega) \approx Ae^{-i\pi/4} \sum_{j=1}^{N} \sqrt{\frac{2}{\pi \rho_j}} \exp\left[i\left(k_0 \rho_j + 2h_j\right)\right],$$
 (7)

where $\rho_j = \rho_1 - (j-1)\Delta\rho$ and $\Delta\rho = d\sin\theta$. The center of the source array is set as reference point and the distance is approximately $\rho = \rho_1 - L\Delta\rho/2d$, where L = (N-1)d is the total length of line source array. Then one has

$$p(\rho, \theta, \omega) \approx A \sqrt{\frac{2}{\pi \rho}} e^{i(k_0 \rho - \pi/4)} \frac{1}{N} \frac{\sin \left[N k_0 d \left(\sin \theta - 2g\right)/2\right]}{\sin \left[k_0 d \left(\sin \theta - 2g\right)/2\right]}.$$
 (8)

Equation (8) implies that the maximum of the radiation always appears at a particular angle, $\theta_0 = \arcsin(2g)$, which is totally independent of k_0 . The field of the extraordinary reflected wave at θ_0 is $p(\rho, \theta_0, \omega) \approx ANe^{i(k_0\rho - \pi/4)}/\sqrt{2/\pi\rho}$. Then, Eq. (8) can be expressed as $p(\rho, \theta, \omega) = p(\rho, \theta_0, \omega)D(\theta)$, where $D(\theta)$, is the directional factor, given as below

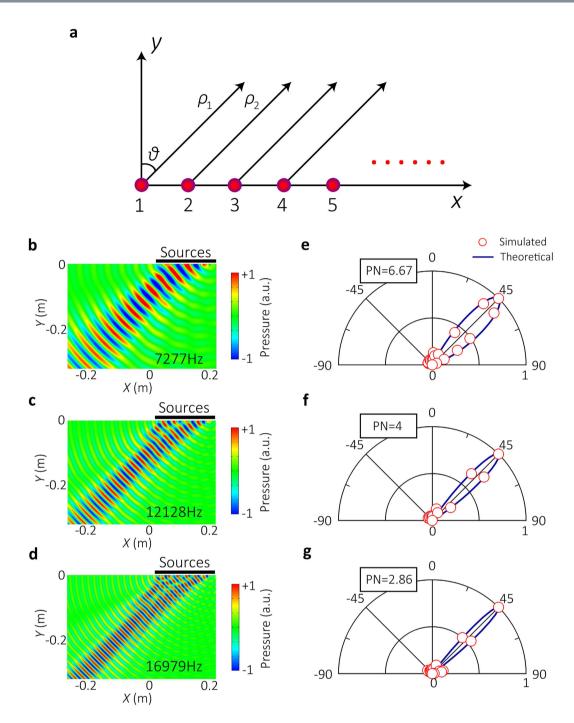


Figure 4. Physical model based on phased array theory. (a) Schematic illustration of the physical model based on phased array theory. The total acoustic pressure field is the sum of the secondary sources with different phase delays. (b, c, d) The acoustic pressure fields given by Eq. (6) at 7.277 kHz, 12.128 kHz, and 16.979 kHz, respectively. (e,f,g) The directional factors $D(\theta)$ given by Eq. (9) (blue line) and the simulated ones (red circle) at 7.277 kHz, 12.128 kHz, and 16.979 kHz, respectively.

$$D(\theta) = \frac{1}{N} \frac{\sin \left[Nk_0 d \left(sin\theta - 2g \right) / 2 \right]}{\sin \left[k_0 d \left(sin\theta - 2g \right) / 2 \right]}.$$
 (9)

Figures 4(b–d) show the acoustic pressure fields given by Eq. (6) at different frequencies. Figures 4(e–g) show a comparison between the values of $D(\theta)$ given by Eq. (9) and by numerical simulation at corresponding frequencies. The theoretical results show that the extraordinary reflections appear at $\theta=45^{\circ}$ without sidelobes, which agree excellently with the numerical and experimental results.

The bandwidth of ERSCS. In principle, our scheme is capable of yielding a completely dispersionless wavefront manipulation. In practice, however, the desired gradient surface must be mimicked by a certain structure, which is necessarily frequency-dependent. Nevertheless, the operating bandwidth should be conveniently adjusted or extended according to the practical requirement. Therefore, it is important to investigate the influence of structural parameters on the efficiency and bandwidth of the ERSCS. We define a parameter of $R = W/W_0$ with W_0 and W being the sound power on the ERSCS and the 'Reflect Area' in Fig. 2(c) respectively, which can be used to quantitatively evaluate the performance of the ERSCS. Figures 5(a, b) show the reflection efficiency spectra simulated for three particular cases: n = 12, 18, and 27 respectively. For the ERSCS model with n = 18, the effective bandwidths, which is defined as R > 0.5, is from 2.5 to 20.1 kHz (about 3 octaves). With a constant period d, as shown in Fig. 5(a), the lower cutoff frequency (f_{cl}) of the ERSCS decreases with increasing n, while the upper cutoff frequency (f_{cl}) remains invariant. On the other hand, with a constant ERSCS width, f_{ch} shifts to higher frequency, while f_{cl} keeps constant when n increases, as shown in Fig. 5(b).

The above results reveal that f_{cl} is determined by the total widths of the ERSCS, whereas f_{ch} is determined by d or PN which can also be written as $PN = c_0/2dgf$, i.e., the resolution of the phase surface. Noting that for different n, f_{ch} is always smaller than the cutoff frequency of the waveguide $f_c = c_0/2d_0$ (e.g., for n = 18, $f_{ch} = 20.1$ KHz and $f_c = 22.9$ kHz). Hence, a sufficiently low discrete resolution will give rise to undesired reflected wave, which determines the value of f_{ch} . In Figs. 5(a,b), upper 3 dB cutoff frequency is approximately PN = 2.41. For an extreme case of PN = 2, which means the phase oscillates between 0 to π , the phase changes are symmetric between x to -x and -x to x, leading to two symmetric try reflected directions with corresponding angles of $\pm 45^{\circ}$, as shown in Figs. 5(c, d). Thus, in Fig. 5(a), the value of R at PN = 2 is nearly half of the value of R at PN > 2.41 because only half of the scattered wave reflects to the desired region. Figures 5(e, f) show that a sidelobe appears at $\theta_0 = -90^\circ$ with PN = 2.41, which corresponds to $k_0d = 3.68$, consistent with the result in Fig. 5(g). This spatial aliasing phenomenon and can be well explained by Eq. (9). $\theta_0 = 45^{\circ}$ is one solution to the equation $D(\theta) = 1$, $k_0 d(\sin\theta - 2g)/2 = 0$. Yet there is another solution, i.e., $k_0 d(\sin \theta$ -2g/ $2 = m\pi$ ($m = \pm 1, \pm 2, \pm 3...$). Generally such a condition is hard to satisfy since the discrete resolution is usually high and the value of $k_0 d(\sin\theta - 2g)/2$ cannot reach $\pm \pi$. When the frequency is high enough (viz., k_0d is sufficiently low), however, another solution corresponding to $k_0d(sin\theta-2g)/2=-\pi$ will appear. Figure 3(g) gives the relationship between the values of K_0d and the extraordinary reflection angles when sidelobe appears at $\theta_0 = -90^{\circ}$, which satisfies

$$k_0 d \left(\sin \theta_r + 1 \right) = 2\pi, \tag{10}$$

where θ_r is the extraordinary reflection angles. The numerical results agree well with the theoretical predictions, in which the value of K_0d gives the upper cutoff frequency of extraordinary reflection. The above analyses suggest that for the purpose of mimicking the desired phase surface by using discrete units which necessarily requires a fine spatial resolution²⁰, the simplified model developed here can be used to predict the upper cutoff frequency accurately. Thanks to the simple structure of the designed subwavelength corrugated surface, by just increasing n one can have extended bandwidth and finer resolution of the phase surface that provides a full freedom for wavefront control that will be applicable to any beam shaping.

Realization of arbitrary dispersionless phase fronts. As we have discussed above, Eq. (3) implies that the general scheme proposed by us allows one to design arbitrary phase profile as a function of x while keeping it frequency-independent. For the particular implementation by the SCS made of subwavelength grooves, such potential could be realized by choosing the gradient of h as a more complicated function of x. Thereby any desired dispersionless phase front can be realized, such as planar focusing, and accelerating non-diffractive beam with arbitrary convex trajectory. In detail, if a spatial phase $\phi(x)$ is needed, according to Eq. (4), the depth of groove should be chosen as

$$h(x) = h_0 - \frac{\phi(x)}{2k_0},$$
 (11)

where h_0 is a constant introduced for ensuring that h(x) is a positive value. Then the resulting phase front is dispersionless due to the fact that $\phi(x)/2k_0$ is independent of wavelength, and the resolution of phase profile can be controlled freely by adjusting the value of n.

Planar focusing. In order to focus sound at an arbitrary position (x_0, y_0) , a hyperboloidal phase profile is required

$$\phi(x) = k_0 \left[\sqrt{(x - x_0)^2 + y_0^2} - \sqrt{x_0^2 + y_0^2} \right].$$
 (12)

According to Eq. (12), the depths of grooves at different x locations are determined by

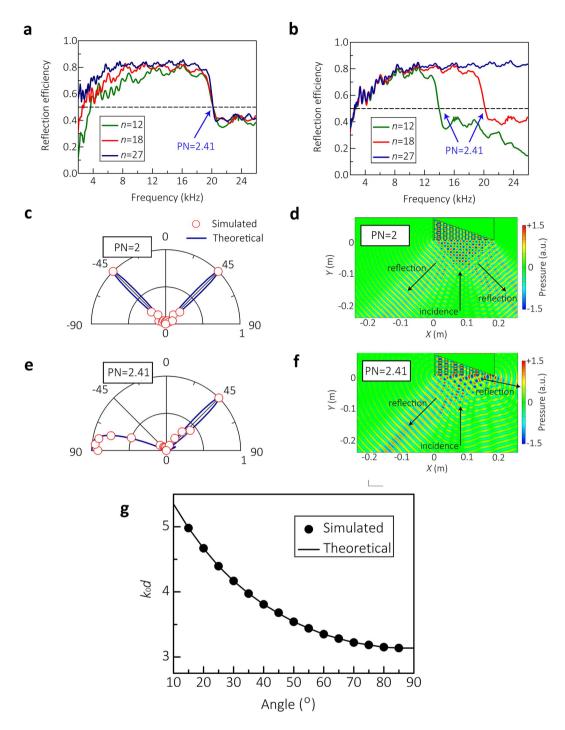


Figure 5. The bandwidth of ERSCS. The reflection efficiency (R) spectra with division numbers (n) of 12, 18, and 27 with the period d being held fixed in (a) and the total width of the ERSCS being constant in (b). (c,e) The direction factors $D(\theta)$ given by Eq. (9) (blue line) and the simulated ones (red circle) at PN = 2 and PN = 2.41 (corresponds to K_0d = 3.68). (d,f) Numerical results of the scattered acoustic pressure field at PN = 2 and PN = 2.41. (g) The relationship between the values of k_0d and the extraordinary reflection angles when the sidelobe appears at $\theta_0 = -90^\circ$.

$$h(x) = h_0 - \frac{1}{2}\sqrt{(x - x_0)^2 + y_0^2}.$$
 (13)

Figures 6(a-c) show the scattered acoustic intensity fields of sound focusing model with the focal location $(10 \, \text{cm}, -19 \, \text{cm})$ at $7.277 \, \text{kHz}$, $12.128 \, \text{kHz}$ and $16.979 \, \text{kHz}$, respectively. The focusing phenomenon is obviously seen in ultra-broadband. The dispersionless focusing phenomenon can be also interpreted

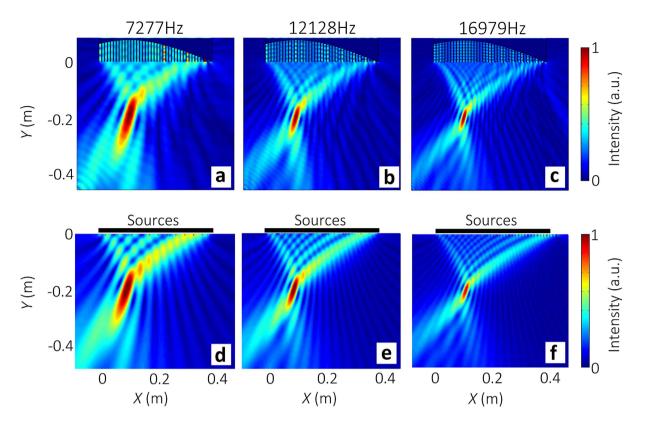


Figure 6. Illustration of sound focusing by SCS. The sample is made of 40 grooves with d=1cm and $d_0=0.75$ cm. (a,b,c) Numerical results of the scattered acoustic intensity fields for normally-incident plane wave at frequencies of 7.277 kHz, 12.128 kHz, and 16.979 kHz, respectively. The focal location is (10 cm, -19 cm). (d, e, f) Acoustic intensity fields given by Eq. (6) at different frequencies.

by phased array theory. Figures 6(d, e) display the acoustic intensity fields given by Eq. (6) based on the phased array theory, which agree with the numerical results. Here, ϕ_j in Eq. (6) is no longer linearly varying with x, but meets Eq. (12).

Non-diffractive beam with arbitrary convex trajectory. We have also demonstrated SCS with generation of acoustic beam that propagates along arbitrary convex trajectory. On the basis of Caustic theory^{21,22}, arbitrary convex accelerating beams can be obtained by directing an appropriate spatial phase profile. The relationship between the phase profile $\phi(x)$ and the reflected angle θ is $d\phi(x)/dx = -k_0 \sin \theta$. To realize an arbitrary trajectory y = f(x), the phase profile can be expressed as²²

$$\frac{d\phi(x)}{dx} = \frac{k_0 f'(y)}{\sqrt{1 + [f'(y)]^2}},$$
(14)

where $f'(y)=\tan\theta$ is the slope of the trajectory. As an example, a circular trajectory $(x+r)^2+y^2=r^2$ with r=0.4m is designed. The slope of the curve at (x_0y_0) is $f'(y_0)=-y_0/\sqrt{r^2-y_0^2}$. Owing to $\tan\theta=(x_0-x)/y_0$ and $(x_0+r)^2+y_0^2=r^2$, the slope of the trajectory can be expressed as

$$\tan \theta = \sqrt{\frac{(x+r)^2}{r^2} - 1} \,. \tag{15}$$

Combining Eq. (14), we obtain

$$\frac{d\phi(x)}{dx} = k_0 \frac{\sqrt{(x+r)^2 - r^2}}{x+r}.$$
 (16)

The spatial phase profile is

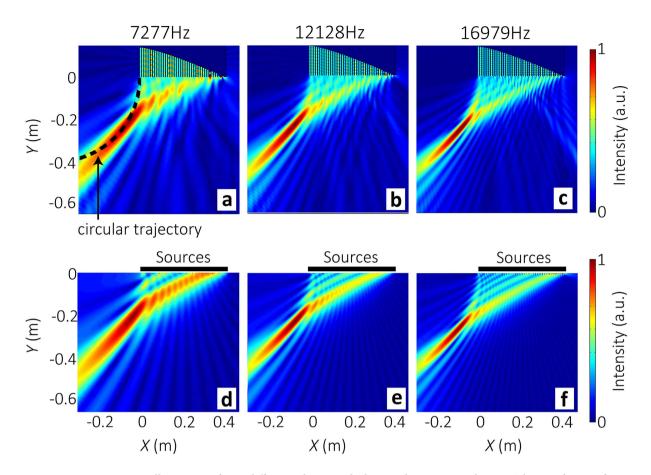


Figure 7. Illustration of non-diffractive beam with the circular trajectory by SCS. The sample is made of 40 grooves with d=1cm and $d_0=0.75$ cm. (a, b, c) Numerical results of the scattered acoustic intensity fields for normally-incident plane wave at frequencies of 7.277 kHz, 12.128 kHz, and 16.979 kHz, respectively. The trajectory is $(x+r)^2+y^2=r^2$ (dashed line). (d, e, f) Acoustic intensity fields given by Eq. (6) at different frequencies.

$$\phi(x) = k_0 \left[\sqrt{(x+r)^2 - r^2} - r \arccos\left(\frac{r}{x+r}\right) \right]. \tag{17}$$

Thus, the depth h(x) is

$$h(x) = h_0 - \frac{1}{2} \left[\sqrt{(x+r)^2 - r^2} - r \arccos\left(\frac{r}{x+r}\right) \right].$$
 (18)

Figures 7(a-c) show the scattered acoustic intensity fields at $7.277\,\mathrm{kHz}$, $12.128\,\mathrm{kHz}$ and $16.979\,\mathrm{kHz}$, respectively. Figures 7(d, e) display the acoustic intensity fields given by Eq. (6), which is in good agreement with the numerical results.

Discussion

Realization of dispersionless manipulation of wavefront is a challenge in optic/acoustic wave steering. We have proposed a general scheme for resolving this challenge and showed that a simple corrugated surface with subwavelength grooves can freely control the reflected waves to form the desired wave field without frequency limitation. An analytical model is developed to interpret this extraordinary phenomenon and can be used to predict the upper cutoff frequency when using a discrete phase array to mimick a continuous phase profile². On the other hand, the proposed reflective structure has the potential to be combined with a waveguide structure for realizing unconventional wave-steering effect in waveguides^{23,24}. Our finding may pave the way to design of ultra-broadband optical/acoustical wavefront-devices, and have potential applications in a variety of practical situations in need of special harness of acoustic wave such as medical ultrasound application or field caustics engineering. The simple configuration and distinct reflective characteristic of the proposed structure also makes it particularly inspiring in the fields of

architectural or environmental acoustics generally associated with the control of reflection of broadband signals, and may result in conceptual devices such as flat walls capable of redirect and reshape sound at will.

Methods

Acoustic measurements. Our experiment was carried in the anechoic chamber in order to eliminate the influence of the reflected waves. An absorptive plate was used to separate the incident and reflected acoustic fields as shown in Fig. 2(c). To obtain an acoustic plane wave, a loudspeaker was located 3 m away from the sample (much larger than the width of the sample). The measuring area and the center of loudspeaker are located in the same *x-y* plane. Two 0.25-inch-diameter Bruel & Kjær type-4961 microphones are used to detect the acoustic pressure, one is used to scan the measuring region, and another is fixed near the loudspeaker to detect the phase information.

Numerical simulations. Numerical simulations are conducted with COMSOL Multiphysics software. The simulated material for ABS has a density 1180 kg/m3 and sound speed 2700 m/s, which are the parameters of the 3D-printed materials in the experiments. The surrounding medium is air with its density 1.21 kg/m³ and sound speed 343 m/s. The viscous effect has been ignored in our simulations, corresponding to the experimental situation, where the thickness of viscous boundary layer, approximating to 0.3 mm for lower frequency limitation, is about 25 times smaller than the groove width.

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Author Contributions

Y.F.Z. and R.Q.L. performed the theoretical simulations; X.Y.Z. and X. J. designed and carried out the experiments; J.C.C. developed the phased array model; Y.F.Z., X.Y.Z., J.T. and J.C.C. wrote the manuscript; B.L. and J.C.C. guided the research. All authors contributed to data analysis and discussions.

Additional Information

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