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# Interfacial Delamination at Multilayer Thin Films in Semiconductor Devices

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**ABSTRACT:** With the evolution of semiconducting industries, thermomechanical failure induced in a multilayered structure with a high aspect ratio during manufacturing and operation has become one of the critical reliability issues. In this work, the effect of thermomechanical stress on the failure of multilayered thin films on Si substrates was studied using analytical calculations and various thermomechanical tests. The residual stress induced during material processing was calculated based on plate bending theory. The calculations enabled the prediction of the weakest region of failure in the thin films. To verify our prediction, additional thermomechanical stress was applied to induce cracking and interfacial delamination by various tests. We assumed that, when accumulated thermomechanical-residual and externally applied mechanical stress becomes larger than a



critical value the thin-film cracking or interfacial delamination will occur. The test results agreed well with the prediction based on the analytical calculation in that the film with maximum tensile residual stress is the most vulnerable to failure. These results will provide useful analytical and experimental prediction tools for the failure of multilayered thin films in the device design stage.

# **1. INTRODUCTION**

Since the invention of field-effect transistors in the 1920s, one of the main issues in the semiconductor industry has been the fabrication of faster and smaller transistors.<sup>1</sup> Currently, for highly integrated devices, the channel width of the device has decreased to less than 10 nm.<sup>2,3</sup> However, the processing technology is limited in terms of further decreasing the channel width.<sup>2,4</sup> Hence, semiconductor devices with multilayered structures have evolved to increase the degree of integration.<sup>4,5</sup> Such multilayered devices have the advantages of high density, low cost, and low power consumption.<sup>3</sup>

For multilayered semiconductor devices, various nanostructures with high aspect ratios, such as nanoplates, nanopillars, and nanowires, should be fabricated.<sup>6</sup> Among the various nanostructures, nanoplates with oxide or nitride/nitride/metal multilayers are one of the essential components in thin-film transistors since oxide, nitride, and metal layers are used as the gate dielectric, adhesion layer, and gate metal in metal/ insulator/semiconductor (MIS) structures, respectively.<sup>4</sup> For smaller channel widths and higher stacks, the thickness of the thin films should be decreased,<sup>7</sup> which makes the thin films more vulnerable to various thermomechanical failures, such as bending, cracking, and, particularly, interfacial delamination.<sup>8,9</sup> These factors eventually lead to the catastrophic failure of the devices.<sup>4</sup>

As stated previously, the interfacial delamination between thin films has become the most critical device failure mode because more complex residual stresses are induced by increasing the number of layers in the multilayered films during deposition processes and operations. Heat release during device operation has increased with the number of layers in the multilayered films due to thermal cycling.<sup>10–13</sup> Upon the cyclic heating and cooling during the deposition process and operation of the devices,<sup>4</sup> thermomechanical-residual stress is induced in the films and substrates due to mismatches in the coefficient of thermal expansion (CTE), Young's modulus, and Poisson's ratio of the constituent film materials and the substrate, which is the well-known cause of failure in thin-film devices.<sup>14</sup>

Simple analytical calculation methods for the residual stresses in the single film deposited on a thick substrate, which can be readily used in industries, have been well developed and can be readily used in industries.<sup>15,16</sup> Hence, the prediction of the weakest region to undergo thermomechanical failure can be readily made if the thermomechanical processing conditions and material properties are known.<sup>17</sup> However, the above-mentioned simple prediction process cannot be applied to the multilayer thin films, and the

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**Figure 1.** (a) Schematic description that thin-film failure occurs beyond a critical stress. (b) Real specimen image and the multilayered structure. "Photograph courtesy of "Hye-Jun Kil." Copyright 2022." (c) Schematic descriptions of strain components in the multilayer system after heating or cooling. (d) Schematic descriptions of the thermomechanical tests performed in this study.

experimental verification process should be carefully designed accordingly.<sup>4,18</sup> Also, comprehensive analytical and experimental methods that could consider the effect of CTE and Young's modulus mismatch in the multilayered system are required.<sup>19</sup> However, current methods that use analytical tools such as finite element modeling to predict fracture behavior by thermal stress used mechanical experiments such as four-point bending tests to verify their prediction.<sup>20,21</sup> The main parameter affecting fracture behavior in these mechanical analyses was mechanical properties such as Young's modulus and Poisson's ratio. Hence, it was unsuitable for thermomechanical analysis in which the CTE mismatch also should be considered.

In this work, the residual stress distribution throughout the multilayer thin films and the substrate induced due to the temperature changes during the deposition was modeled analytically based on plate bending theory.<sup>18,22</sup> Plate bending theory has been widely used to model the residual stress distribution in multilayered structures.<sup>23,24</sup> The detailed

structures and material properties of two multilayer specimens are summarized in Figure 1 and Tables 1 and 2.

## Table 1. Descriptions of Specimens A and B

	Si	SiO <sub>2</sub>	SiN	TiN	W
specimen A	770 $\mu$ m	100 nm		15 nm	40 nm
specimen B	770 $\mu$ m		100 nm	15 nm	40 nm

#### Table 2. Properties of the Materials Used in This Work

	(10-6/177)	
materials	$\alpha$ (10 °/K)	E (GPa)
Si	2.6 <sup>25</sup>	130 <sup>25</sup>
SiO <sub>2</sub>	0.7 <sup>26</sup>	70 <sup>27</sup>
SiN	3.27 <sup>28</sup>	149 <sup>29</sup>
TiN	9.4 <sup>30</sup>	250 <sup>31</sup>
W	4.3 <sup>32</sup>	410 <sup>33</sup>

We focused on the external stresses such as significant temperature change during the operation as the main reason for the device failure, because the device did not show any failure after the deposition process. Hence, we assumed that the residual stresses during deposition processes were much lower than the critical stress for the device failure, as schematically shown in Figure 1a. The multilayered structures were deposited based on optimized processes to minimize defects, and residual stress occurred during fabrication processes.

The experimental investigation of the failure modes was simultaneously performed by four different thermomechanical test methods, including thermal annealing, laser irradiation, scratch test, and four-point bending test. Especially, the four-point bending test has been frequently used for calculating interfacial fracture energy in various thin-film structures.<sup>21</sup> To identify the location of delamination, analyses by field-emission scanning electron microscopy (FE-SEM) and transmission electron microscopy (TEM) were performed. The experimental results clearly verified that the tensile residual stresses induced at the film are the primary factor for the interfacial delamination and cracking.

# 2. ANALYTICAL CALCULATION AND EXPERIMENTAL PROCEDURES

All of the experiments were carried out using multilayered specimens with thin films deposited in the order of 100 nm silicon oxide (SiO<sub>2</sub>) or silicon nitride (SiN), 15 nm titanium nitride (TiN), and 40 nm tungsten (W) on a 770  $\mu$ m thick silicon (Si) wafer. A schematic description of the samples is shown in Figure 1a,b and Table 1. First, 100 nm SiO<sub>2</sub> or SiN was deposited on a Si wafer using chemical vapor deposition (CVD), then, 15 nm TiN was deposited using atomic layer deposition (ALD), and finally, 40 nm W was deposited using CVD. The largest temperature change ( $\Delta T$ ) during deposition processes is 490 °C.

The residual stress distribution in the multilayers during the deposition process was calculated using plate bending theory.<sup>18,22</sup> The distribution of thermomechanical-residual stress within the multilayered thin films was theoretically predictable through the following calculations. Strain due to heating and cooling of a multilayer system with n thin film is<sup>22</sup>

$$\varepsilon = c + \frac{z - t_{\rm b}}{r} \quad (-t_{\rm s} \le z \le h_n) \tag{1}$$

where *c* is the uniform strain component, *z* is the bending axis,  $t_b$  is the bending strain component, 1/r is the curvature of the sample,  $t_s$  is the thickness of the Si wafer, and  $h_n$  is the thickness of the entire *n* films. Stress distribution on the Si wafer substrate is calculated as<sup>22</sup>

$$\sigma_{\rm s} = E_{\rm s}(\varepsilon - \alpha_{\rm s}\Delta T) \quad (-t_{\rm s} \le z \le 0) \tag{2}$$

where  $E_s$  is Young's modulus of the Si wafer,  $\alpha_s$  is the CTE of the Si wafer, and  $\Delta T$  is the temperature difference during heating or cooling. Stress distribution on each thin film (*i*th from the interface with the substrate) can be expressed as<sup>22</sup>

$$\sigma_i = E_i(\varepsilon - \alpha_i \Delta T) \quad (i = 1 \text{ to } n)$$
(3)

where  $E_i$  is Young's modulus and  $\alpha_i$  is the CTE of the *i*th thin film. Based on the above equation, a general solution can be obtained. The uniform strain component (eq 5) and bending strain component (eq 7) can be obtained from the force

balance equation (eq 6), and the curvature (eq 9) can be obtained from the momentum balance equation (eq 8).<sup>22</sup>

$$E_{\rm s}(c-\alpha_{\rm s}\Delta T)t_{\rm s} + \sum_{i=1}^{n} E_{i}(e-\alpha_{i}\Delta T)t_{i} = 0$$
(4)

$$c = \frac{(E_s t_s \alpha_s + \sum_{i=1}^{n} E_i t_i \alpha_i) \Delta T}{E_s t_s + \sum_{i=1}^{n} E_i t_i}$$
(5)

$$\int_{-t_{\rm s}}^{0} \frac{E_{\rm s}(z-t_{\rm b})}{r} \,\mathrm{d}z + \sum_{i=1}^{n} \int_{h_{i-1}}^{h_{i}} \frac{E_{i}(z-t_{\rm b})}{r} \,\mathrm{d}z = 0 \tag{6}$$

$$t_{\rm b} = \frac{-E_{\rm s}t_{\rm s}^{\ 2} + \sum_{i=1}^{n} E_{i}t_{i}(2h_{i-1} + t_{i})}{2(E_{\rm s}t_{\rm s} + \sum_{i=1}^{n} E_{i}t_{i})}$$
(7)

$$\int_{-t_{\rm s}}^{0} \sigma_{\rm s}(z-t_{\rm b}) \, \mathrm{d}z + \sum_{i=1}^{n} \int_{h_{i-1}}^{h_{i}} \sigma_{i}(z-t_{\rm b}) \, \mathrm{d}z = 0 \tag{8}$$

$$\frac{1}{r} = \left\{ 3 \left[ E_{s}(c - \alpha_{s} \Delta T) t_{s}^{2} - \sum_{i=1}^{n} E_{i} t_{i}(c - \alpha_{i} \Delta T) (2h_{i-1} + t_{i}) \right] \right\} / \left\{ E_{s} t_{s}^{2} (2t_{s} + 3t_{b}) + \sum_{i=1}^{n} E_{i} t_{i} [6h_{i-1}^{2} + 6h_{i-1} t_{i} + 2t_{i}^{2} - 3t_{b} (2h_{i-1} + t_{i})] \right\}$$

$$(9)$$

Since  $t_s$  is much thicker than the thickness of the thin film, the first-order approximation can be applied,<sup>22</sup> and eqs 5, 7, and 9 are simplified as eqs 10–12, respectively. Finally,  $\sigma_s$  and  $\sigma_i$  can be calculated as in eqs 13 and 14, respectively

$$c = \alpha_{\rm s} \Delta T + \sum_{i=1}^{n} \frac{E_i t_i (\alpha_i - \alpha_{\rm s}) \Delta T}{E_{\rm s} t_{\rm s}}$$
(10)

$$t_{\rm b} = -\frac{t_{\rm s}}{2} \left[ 1 - \sum_{i=1}^{n} \frac{E_i t_i}{E_{\rm s} t_{\rm s}} \right]$$
(11)

$$\frac{1}{r} = 6 \sum_{i=1}^{n} \frac{E_i t_i (\alpha_i - \alpha_s) \Delta T}{E_s {t_s}^2} = \sum_{i=1}^{n} \frac{1}{r_i}$$
(12)

$$\sigma_{s} = \frac{2}{t_{s}^{2}} \left( 3z + 2t_{s} - \frac{2}{E_{s}} \sum_{j=1}^{n} E_{j} t_{j} \right) \sum_{i=1}^{n} E_{i} t_{i} (\alpha_{i} - \alpha_{s}) \Delta T$$
$$(-t_{s} \leq z \leq 0)$$
(13)

where  $t_s$ ,  $t_j$ ,  $t_i$ ,  $E_s$ ,  $E_{ij}$ ,  $E_i$ ,  $\alpha_s$ ,  $\alpha_{ij}$  and  $\Delta T$  are the thicknesses of the Si substrate, the *j*th thin film, and the *i*th thin film; Young's modulus values of the Si substrate, the *i*th thin film, and the *j*th thin film; CTE of the Si substrate and the *i*th thin film; and the temperature change during post thermal annealing, respectively.

$$\sigma_{i} = E_{i} \left[ \alpha_{s} - \alpha_{i} + 4 \sum_{j=1}^{n} \frac{E_{j} t_{j} (\alpha_{j} - \alpha_{s})}{E_{s} t_{s}} \right] \Delta T$$
(14)

To verify our calculation result based on eq 14, we applied the well-known assumption that there is a critical stress over which a material fails.<sup>34</sup> If the critical stress is known, and we can measure the external stress initiating the failure, the



Figure 2. Cross sectional TEM images of (a)  $SiO_2^{-}$  and (b) SiN-based multilayered systems. Stress distribution in the Si substrate of the (c) Specimen A and (d) Specimen B. Stress distributions in the deposited thin films of (e) Specimen A and (f) Specimen B. Schematic stress distributions of (g) Specimen A and (h) Specimen B after cooling. A Pt layer shown in Figure 2a,b was deposited for TEM analysis, which is irrelevant to the multilayered systems.

residual stress can be calculated (Figure 1a). Herein, we focused on the situation in which failure occurs as the device without failure operates. That is, the residual stress is a level that does not reach the critical stress, and a value equal to or higher than the threshold stress is applied by the external stress generated while the device is operating, thus causing failure. A specimen having a very low defect density through an optimized manufacturing process could be used to implement a state in which there was no failure before operating.

In this study, external stress was imposed on the samples by four different methods: thermal annealing and quenching, laser irradiation, scratching, and four-point bending tests. For the thermal annealing and quenching test, the specimens in Table 1 were put into a thermal furnace at 800 °C for 75 s. Since the annealing time was quite long, we could ignore the thermal diffusion throughout the specimen. Then, the annealed samples were quenched to room temperature using an aluminum heat sink. As  $\Delta T$  in the thermal test is larger than that of the deposition process (490 °C), greater residual stresses than those during the deposition were induced in the thin films.

For laser irradiation tests, a pulsed ND:YAG laser with a wavelength of 1064 nm and a pulsed width of 6 ns was used. For irradiation, one shot of the laser beam was irradiated on



Figure 3. Field-emission scanning electron microscopy (FE-SEM) images of the delaminated surfaces of (a), (b)  $Si/SiO_2/TiN/W$  and (c), (d) Si/SiN/TiN/W induced by thermal annealing ((a) and (c)) and laser irradiation ((b) and (d)). The red rectangular area in Figure 3a-d is the location where cross sectional sampling for TEM was performed using FIB. Cross sectional TEM and EDS mapping images of the (e)  $Si/SiO_2/TiN/W$  system and (g) Si/SiN/TiN/W system after thermal annealing and quenching. Cross sectional TEM and EDS mapping images of the (f)  $Si/SiO_2/TiN/W$  system and (h) Si/SiN/TiN/W system after laser irradiation.

the specimen surface. By the irradiation,  $\Delta T$  is expected to be around 9500 °C.<sup>35</sup> The heating and cooling cycle will be significantly shorter than the thermal test. Since the pulse duration of the laser was relatively long, and the thickness of each layer was less than 100 nm, we could assume that the absorption length of the laser energy in the multilayered structure was smaller than the thickness of each layer. Hence, thermal diffusion was not considered in the laser irradiation analysis.<sup>36,37</sup> During the laser irradiation, external stresses due to significant temperature change overcome the critical stress, resulting in the failure of the multilayered structures.

A scratch test was conducted in accordance with ISO 20502 and ASTM C1624.<sup>38</sup> A nanoscale scratch tester (CSM, USA) with a sharp diamond tip was used. The diamond tip is 2  $\mu$ m in

radius. The vertical load at the start is 0.5 mN, and the end force is 120 mN. During the scratch test, the diamond tip scratches the surface of the specimen with a constant increase in the vertical load at a rate of 2 mN/s. Hence, the adhesion force of the thin film could be verified by measuring the vertical loads where delamination occurs.

For the four-point bending test, specimens were cut into 50 mm by 8 mm rectangles using a laser dicing machine. Two specimens were bonded together using an epoxy, with the thin films facing each other (Figure 1d). Additionally, in one of the bonded specimens, a notch was formed on the opposite side of the deposited thin films using a dicing saw. The four-point bending test was performed using a universal testing machine (CT-UTM101E, CoreTech, Korea). The displacement veloc-



Figure 4. OM surface images of the (a) Specimen A and (b) specimen B after the scratch test. Cross sectional TEM and EDS mapping images of the (c) Specimen A and (d) Specimen B after scratch tests.

ity was 10 mm/min, and the inner pin and outer pin distances were 20 and 45 mm, respectively (Figure 1d). During the fourpoint bending test, tensile stress was applied to the specimen. The crack initiated at the notch propagated through the layers. When the crack reached the interface with weak interfacial adhesion strength, the crack would propagate through the interface. Otherwise, the crack would propagate throughout the layers further, which would result in surface crack propagation.<sup>21</sup> Hence, in the four-point bending test, the interfacial adhesion strength is critical for determining the crack propagation path.

The cross sections of the failed specimens were analyzed using FE-SEM (JSM-6701F, JEOL, Japan), and the crack location and site of delamination were identified. The delaminated or cracked area was milled using a focused ion beam (FIB, crossbeam 540, ZEISS, Germany), and, using TEM (JEM-F200, JEOL, Japan), materials were identified with energy-dispersive X-ray spectroscopy (EDS) mapping.

#### 3. RESULTS AND DISCUSSION

**3.1. Calculation of the Residual Thermal Stress Based on Plate Bending Theory.** The cross sectional TEM images of Specimen A and B are shown in Figure 2a,b. TiN was deposited between  $SiO_2$  or SiN and W to improve adhesion. As shown in Figure 2a,b, defects and blisters were not observed after deposition. The properties of the materials used in this work are summarized in Table 2. To calculate the residual

stress induced in the multilayered systems during the deposition process, plate bending theory was used, as described in detail in Section 2.<sup>18,22</sup> Based on plate bending theory, the residual stress induced in the Si substrate ( $\sigma_s$ ) with n thin films is expressed in eq 13. The thermal residual stress induced in the *i*th thin film from the interface with the Si substrate is expressed in eq 14.

In Specimens A and B in Figure 2a,b, the thermomechanicalresidual stresses were calculated using eqs 13 and 14, and the results are plotted in Figure 2c–f. As shown in Figure 2c,d, a neutral plane is located at 513  $\mu$ m from the surface of the Si substrate both in Specimens A and B, which is approximately 2/3 of the total thickness of the Si substrate.

As shown in Figure 2e,f, the residual stresses in the thin films are much higher than those in the Si substrates. In the thin films, the sign of the stress is determined by the CTE mismatches between the Si substrate and the thin films. CTE of SiO<sub>2</sub> is lower than that of the Si substrate; hence, compressive stress is generated in the SiO<sub>2</sub> layer (Table 2). On the other hand, tensile stress is induced in the SiN, TiN, and W layers since their CTEs are higher than that of the Si substrate (Table 2). According to the stress distribution in the thin films shown in Figure 2e,f, the strain mismatch is largest at the interfaces between SiO<sub>2</sub> and TiN and between SiN and TiN for Specimen A and B, respectively. Based on the stress distributions, TiN where the largest tensile stress is induced and for which the stiffness is the lowest due to the smallest thickness seems to be the most vulnerable layer to cracking and delamination.

3.2. Thermal Annealing and Laser Irradiation. To verify the prediction made in Section 3.1, additional thermomechanical stresses were applied by thermal annealing and laser irradiation. First, samples were thermally annealed at 800 °C and quenched to room temperature. This  $\Delta T$  is larger by 375 °C than that induced during deposition. An Al heat sink was used to quench the heated samples, and an IR camera was used to confirm that the temperature was immediately reduced to room temperature. As shown in the FE-SEM images in Figure 3a,c, a delaminated area was observed on the multilayered samples. According to the cross sectional TEM images and EDS mapping results in Figure 3e,g, TiN layers were delaminated or cracked under the thermomechanical stresses in both Specimens A and B, respectively. This agrees very well with the theoretical prediction made in Section 3.1. The void formation shown in Figure 3g is that quenching was performed in the atmosphere after thermal annealing. The difference between Young's modulus and Poisson's ratio causes thermal expansion mismatch between layers, resulting in delamination. At this point, microcracks occurred, and voids were created between TiN and SiN.35

In the laser irradiation test (Figure 1d), the laser heat was focused on a small area of specimen surfaces for a very short time, and the specimens were cooled in air.<sup>40</sup> First, the critical pulse energy for delamination was investigated by observing the irradiated surface using an optical microscope (OM). As shown in Figure S1 (Supporting Information), when the pulse energies were 57 and 90 mJ, delamination and cracking occurred on the surfaces of Specimens A and B, respectively. When the laser power was higher than this critical power, the total delaminated area increased, as shown in Figure S1. The detailed delaminated surface at the critical laser power is shown in Figure 3b,d. The main reason for the delamination or cracking of the specimens' surface by this laser irradiation is the difference between Young's modulus and Poisson's ratio between the multilayers. The difference between the two values causes a difference in the degree of expansion of the layers, and it causes the occurrence of strain by this mismatch, occurring delamination. Based on the cross sectional images shown in Figure 3f,h, delamination occurred at the interfaces between SiO<sub>2</sub> and TiN and between SiN and TiN, respectively, which corresponds very well with the thermal annealing tests as well as the calculation result in Section 3.1.

**3.3. Nanoscratch and Four-Point Bending Tests.** Additional mechanical stresses were imposed on the two specimens by nanoscratch and four-point bending tests to verify the relationship between the residual stresses and the thin-film failure. The mean critical forces of the tangential force at which delamination occurred were measured as 7 mN for Specimen A and 15 mN for Specimen B. As in the laser irradiation test, a larger energy was required to delaminate Specimen B than Specimen A. Also, the delaminated films by the nanoscratch test were TiN as in the thermal annealing and laser irradiation tests. The interfacial delamination at SiO<sub>2</sub>/TiN in Specimen A and SiN/TiN interfaces in Specimen B was confirmed by OM, TEM images, and EDS mapping, as shown in Figure 4.

Finally, the four-point bending test was performed to further verify the layer that is most vulnerable to failure in the multilayered systems. A notch was formed at the bottom of the Si wafer, as shown in Figure 5a, and two samples were bonded



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**Figure 5.** (a) OM image of the notch induced on the opposite surface of the thin-film-deposited Si wafer. "Photograph courtesy of "Jin-Hoon Kim." Copyright 2022." (b) FE-SEM image of the Si/SiN/ TiN/W system after the four-point bending test. The yellow dotted line represents the point where the cohesive failure of the epoxy occurred.

using epoxy as in Figure 1d. When the mechanical bending was induced, a crack was generated at the notch, and it was propagated to the bonded interfaces. Then, the crack propagated along the interface or through the layer if the summation of the mechanical stress and the residual stress becomes larger than a critical value for failure.<sup>17</sup>

First, the cross sections of the delaminated surface after the four-point bending test were analyzed. As shown in Figure 6a,c, delamination occurred at the interface between SiO<sub>2</sub> and TiN in Specimen A, which matches well with the previous test and calculation results in Sections 2 and 3, respectively. In Specimen B, interfacial delamination was observed at the interface between SiN and TiN, and the interfacial crack detoured through the W layer rapidly. As shown in Figure 6a, the branching of cracks was not observed in specimen A because cracks propagated through the weak SiO<sub>2</sub>/TiN interface, as shown in Figure 6c. Whereas, as shown in Figure 6b, branching of cracks was observed on the surface of Specimen B because propagated cracks detoured through the W layer.

Based on fracture mechanics, if the interfacial strength was stronger than the intrinsic material's strength, the crack propagated through the thin film in the vertical direction. Whereas when the interfacial strength was weaker than the intrinsic material's strength, the crack propagated through the weak interface in the lateral direction.<sup>41</sup> Crack branching occurred due to rapid rupture of the brittle materials. In specimen B, since the crack propagated through films (vertical direction to the specimen), the crack rapidly reached the surface. Hence, the branching was observed, as shown in Figure 6b. Whereas in specimen A, since the crack propagated through the interface (lateral direction), it took longer for cracks to reach the surface. Hence, we could not observe the branching in specimen A, as shown in Figure 6a. Also, as shown in Figure 6e, we could observe a plateau in the stressdisplacement curve, which indicates slow crack propagation through the interface. Such differences in crack propagation behavior observed after four-point bending tests strongly indicate that the SiN/TiN interface has stronger adhesion strength compared with SiO<sub>2</sub>/TiN, also shown in the analytical calculation, laser irradiation, and scratch tests.

The interfacial fracture energy, which is the critical strain energy release rate (*G*), can be calculated from the force–displacement curve obtained from the four-point bending test, which is expressed as<sup>42</sup>

$$G = \frac{21(1-\nu^2)M^2}{4Eh^3}$$
(15)



Figure 6. Surface and cross sectional FE-SEM images of the (a)  $Si/SiO_2/TiN/W$  system and (b) Si/SiN/TiN/W system after the four-point bending test. Cross sectional TEM and EDS mapping images of the (c)  $Si/SiO_2/TiN/W$  system and (d) Si/SiN/TiN/W system after the four-point bending test, respectively. (e) Force-displacement curves of the  $Si/SiO_2/TiN/W$  system obtained during the four-point bending test.

where *M* is the bending moment, which can be expressed as M = PL/2b in the plate beam; *P* is the bending load; *L* is the spacing between loading pins; *b* is the width of the specimen;  $\nu$  is Poisson's ratio of the Si substrate; *E* is Young's modulus of the Si substrate; and *h* is the thickness of the specimen.

The force-displacement curve obtained from the four-point bending test is shown in Figure 6e, revealing that the force was linearly increased first, which is the overall deformation of the specimen. Then, a crack was formed at the notch and propagated through the bonded interface. The crack at the notch led to the formation of a spike in the forcedisplacement curve, as shown in Figure 6e. After the spike was formed, the load was relaxed and then showed linearelastic behavior. When the displacement increased further, a plateau was formed, where delamination started. The force value at the plateau was used to calculate G using eq 15, and the calculated G value was 19  $J/m^2$  in Specimen A, which is comparable to the reported G values of thin films deposited on the Si substrate.<sup>43–45</sup> After the crack was formed at the notch, the bending of the specimen mostly occurred to the unnotched side of the specimen (schematically shown in Figure

1d), which is independent of the notch's depth. Hence, in the four-point bending test, the effect of notch depth is known to be negligible.<sup>46</sup>

Kim et al. measured the *G* of a photosensitive dielectric/Ti/ Cu multilayered system deposited on a Si wafer using the fourpoint bending test, and they reported that interfacial delamination occurred at the photosensitive dielectric/Ti interface; the *G* value was approximately 17.3 J/m<sup>2.43</sup> Völker et al. measured the *G* value of a borophosphosilicate dielectric glass (BPSG)/Ti/W/AlSiCu multilayered system on a Si wafer using the four-point bending test.<sup>44</sup> They also reported that delamination occurred at the BPSG/Ti interface, and the *G* value was approximately 11.7 J/m<sup>2.44</sup> Ma et al. analyzed the adhesion property of a Si/SiO<sub>2</sub>/AlCu/TiN/SiO<sub>2</sub> multilayered system on a Si wafer using the four-point bending test. They also reported that delamination occurred at the SiO<sub>2</sub>/TiN interface, but the interfacial fracture energy was not measured.<sup>45</sup>

### 4. CONCLUSIONS

In this study, the film failure that occurred in oxide/nitride/ metal or nitride/nitride/metal multilayer systems for semiconductor devices was analyzed theoretically and experimentally. Based on plate bending theory, the largest tensile residual stress was predicted to be induced at the TiN layer in two different specimens, which makes the TiN layer and its interfaces with TiN most vulnerable to cracking and delamination, respectively. To verify the theoretical prediction, thermal annealing and laser irradiation tests were performed to induce additional thermomechanical stresses. In addition, external mechanical stress was also induced using the scratch test and four-point bending test. The four test results all agreed well with the theoretical prediction that the TiN layer and the interfaces with it are the weakest region to failure. These agreements clearly confirm that the region of failure in multilayered thin films can be predicted based on the residual stresses that are calculated analytically. In conclusion, the analytical model and experimental test methods proposed in this work will be useful tools to evaluate the mechanical stability of a designed multilayer thin-film system on Si before fabrication and during operation.

# ASSOCIATED CONTENT

#### **Supporting Information**

The Supporting Information is available free of charge at https://pubs.acs.org/doi/10.1021/acsomega.2c02122.

Photographs and optical microscope (OM) images of (a) specimen A and (b) specimen B with different laser irradiation power (PDF)

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## **Author Contributions**

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# Notes

The authors declare no competing financial interest.

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