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Quasi-Static Pull-in: an Instability in Electrostatic Actuators

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We identify a new instability in electrostatic actuators dubbed quasi-static pull-in. We report experimental evidence of the instability and study its characteristics in two types of micro actuators operating in ambient air. We found that the underlying mechanism is a fast-slow dynamic interaction between slowly-varying electrostatic excitation and fast resonator response that instigate large non-resonant oscillatory orbits and eventually disappears in a global Shilnikov bifurcation. Based on these findings, we formulate and present a new taxonomy of pull-in instabilities in electrostatic actuators.

Nonlinearities in electrostatic Micro-Electro-Mechanical Systems (MEMS) are a rich source of interesting dynamic phenomena. Sources of nonlinearity in electrostatic MEMS include the dependence of the electrostatic force on displacement, geometric and inertial nonlinearities, nonlinear damping mechanisms, and interactions with the substrate¹. They result in static and dynamic bifurcations, multivaluedness, and chaos. These phenomena have been exploited to design high sensitivity sensors, large amplitude actuators, mechanical memory bits, and encryption keys^{2–6}.

One of the most important nonlinear phenomena in electrostatic MEMS is the pull-in instability⁷ where the moving structure snaps to the actuation electrode. While significant efforts have been devoted to study this phenomenon, a consistent taxonomy of its different types and underlying mechanisms is yet to emerge. We posit that a classification system based on the ratio of frequency f of the excitation to the actuator's fundamental natural frequency f_n can achieve that.

Static pull-in is measured using quasi-static ramp waveforms, ($ff_n \rightarrow 0$), which minimize inertial and damping effects. As the voltage between the actuator and an electrode increases monotonically, the stable equilibrium (node) and unstable equilibrium (saddle) coincide at a saddle-node bifurcation. Beyond this point, the actuator snaps to the electrode. Krylov and Maimon⁸ and Khater *et al.*³ utilized ramp waveforms with frequencies of $f = 1$ kHz and $f = 1.8$ Hz, respectively, to measure static pull-in. While a step or other waveforms may also be used, the common characteristic of static pull-in is transient (non-repeatable) dynamics.

A margin of stability exists around the saddle-node bifurcation. The size of this margin is proportional to the waveform rise time and the actuator quality factor⁹. A slow rise allows for accurate determination of the bifurcation point corresponding to static pull-in voltage. A fast rise instigates transients, thereby decreasing the effective static pull-in voltage. Many researchers have investigated the boundaries of this margin^{8,10,11}.

Dynamic pull-in is instigated by resonant waveforms where the excitation frequency and one of the natural frequencies are integer multiples or submultiples of each other ($ff_n \approx p/q$), such as the case for primary, superharmonic, and subharmonic resonances^{12,13}. Dynamic pull-in occurs at lower RMS voltage than static pull-in because of the dynamic amplification available at resonance¹⁴. In low damping regimes, it occurs subsequent to a homoclinic tangle resulting in erosion of the safe motion basin^{13,15,16} when it reaches about 50% of the basin size¹⁷. It may also occur subsequent to a cyclic-fold bifurcation^{13,17,18}. In high damping regimes, it occurs due to a homoclinic bifurcation as an orbit touches the stable manifold of the saddle¹⁶. Further, in all cases a margin of stability exists around the stable manifold of the saddle. Transients around a stable orbit that cause the actuator to wander beyond this margin also lead to dynamic pull-in¹⁴.

In this work, a new class of pull-in instabilities, dubbed quasi-static pull-in, is observed in orbital (repeatable) non-resonant dynamics where the excitation frequency is finite but much smaller than the fundamental natural frequency, $ff_1 < 1$. It was observed in two actuators excited by biased AC voltage in ambient air. Table 1 summarizes the conditions and mechanisms of those instabilities. This paper reports on the identification of this instability and characterization of its underlying mechanism: a global Shilnikov bifurcation¹⁹.

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Type	Frequency Ratio	Mechanism
Static	$ff_1 \rightarrow 0$	Saddle-node bifurcation
Quasi-static	$ff_1 \ll 1$	Shilnikov bifurcation
Dynamic	$ff_n \propto p/q$	(i) homoclinic bifurcation
		(ii) homoclinic tangle
		(iii) cyclic-fold bifurcation

Table 1. A taxonomy for pull-in instabilities.

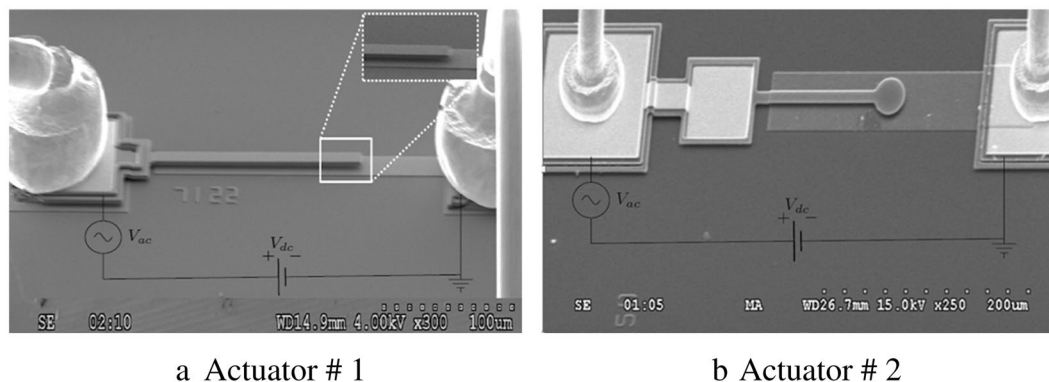


Figure 1. Scanning Electron Microscopy (SEM) pictures of the actuators (1) and (2) which consist of cantilever beam and cantilever beam attached with a plate at the end, respectively.

Shilnikov bifurcation is characterized by the appearance of an orbit homoclinic to a saddle focus relevant to our case arises due to the presence of an unstable equilibrium (a saddle) characterized by the three eigenvalues closest to the imaginary axis having the form: $(-\rho \pm i\omega, \lambda)$ where $\lambda, \rho > 0$ ^{19,20}. The orbit spirals towards the saddle then departs away from it in a large excursion along its unstable manifold. The ratio of the real part of the complex pair of eigenvalues to the real eigenvalue $\delta = \rho/\lambda$ is called the saddle index. When $\delta < 1$ Shilnikov homoclinic orbits appear and Shilnikov chaos exists in their neighborhood^{19,20}.

Shilnikov homoclinic orbits and Shilnikov chaos were previously observed in feedback controlled single mode laser²¹, intracavity multimode solid-state laser²², DC voltage excited plasma at a low gas pressure²³, and fluid flows in concentric rotating cylinders²⁴. In this paper, we report observation of Shilnikov orbits homoclinic to a saddle focus as well as Shilnikov chaos in electrostatic MEMS actuators undergoing quasi-static pull-in.

Methods

Experiments were conducted on two types of electrostatic MEMS actuators vibrating in their first out-of-plane bending mode. Actuator # 1, Fig. 1(a), is a microcantilever beam, whereas actuator # 2, Fig. 1(b), has a circular plate at the end of the beam. Substrate electrodes provide electrostatic actuation.

The actuators were fabricated using PolyMUMPs fabrication process²⁵. The beams were fabricated in Poly 1 structural layer with dimensions of $(175 \times 10 \times 2 \mu\text{m}^3)$ for actuator # 1 and $(115 \times 10 \times 2 \mu\text{m}^3)$ for actuator # 2. Substrate electrodes were patterned in Poly0 layer under the full beam length. The capacitive gap was experimentally found to be $d = 2 \mu\text{m}$ for both actuators. Two gold pads were patterned at the root of the beam and the end of the bottom electrode to apply potential difference to the actuator.

Experiments were conducted in atmospheric pressure while the actuators were placed inside a metal enclosure to protect against stray magnetic fields. A function generator was used to supply the actuation waveform. A laser-Doppler vibrometer²⁶ was utilized to measure the actuator tip velocity and displacement. The measurements were digitized using a digital oscilloscope.

Results

The frequency-response curves of the tip velocity, shown in Fig. 2, were obtained under constant voltage waveforms to characterize the actuators response over a wide frequency range. Each curve is composed of a forward and a backward frequency sweep. The voltage waveform and frequency range were $V_{dc} = V_{ac} = 7.725 \text{ V}$ and $f = [5-90] \text{ kHz}$ for actuator # 1 and $V_{dc} = V_{ac} = 7.125 \text{ V}$ and $f = [5-60] \text{ kHz}$ for actuator # 2. The slew rate was set to 2.5 kHz/s to minimize transient effects. Data was collected in time windows of 0.4 s at a sampling frequency of $f_s = 313 \text{ kHz}$. The time-domain data was post-processed to obtain the RMS velocity of the tip over a time window of 20 excitation periods (T) and assigned to the frequency value at the window mid-point.

For both actuators, no motions were observed for excitation frequencies lower than 5 kHz with the actuator going into and remaining in contact with the substrate throughout the excitation cycle. Periodic and aperiodic motions appeared for excitation frequencies $f \geq 5 \text{ kHz}$. The positive slope lines observed in the frequency ranges $f = [5.0-6.490] \text{ kHz}$ and $f = [53.0-70.0] \text{ kHz}$ for actuator # 1 and $f = [5.0-10.627] \text{ kHz}$ for actuator # 2 are evidence

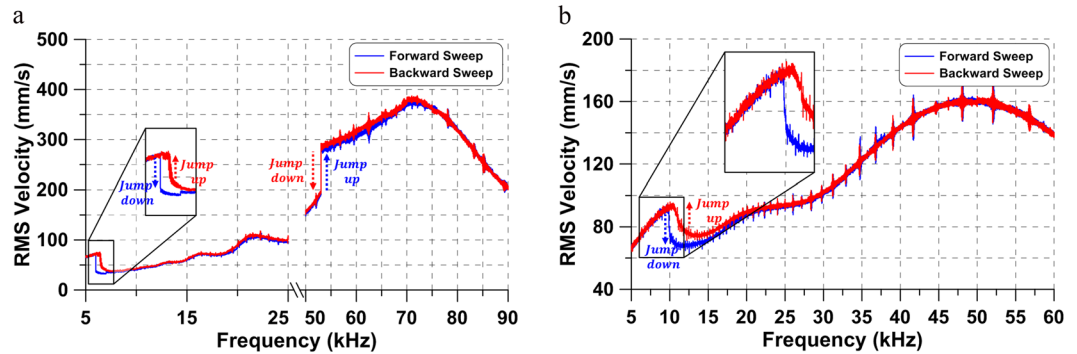


Figure 2. The frequency-response curves of (a) actuator #1 under the excitation voltage $V_{dc} = V_{ac} = 7.725$ V and (b) actuator #2 under the excitation voltage $V_{dc} = V_{ac} = 7.125$ V. Forward and backward frequency-sweeps are colored in blue and red, respectively.

of tapping-mode oscillations where the actuator tip comes into regular contact with the substrate. In these ranges, the limiting effect of the substrate maintains the tip displacement almost constant at a value close to the gap distance $d = 2 \mu\text{m}$. As a result, the velocity frequency-response varies almost linearly with frequency.

The tapping-mode oscillations observed at low-frequency are large even though they occur faraway from primary resonance $f/f_1 < 1$ and its superharmonics. In fact, while actuator #1 also experiences similar sized tapping mode oscillations and dynamic pull-in at primary ($f \approx f_1$) resonance, actuator #2 does not experience the large orbits, leading to tapping or pull-in, except in that low frequency range. Further, the low-frequency tapping-mode orbits jump-down during forward frequency sweeps to a branch of smaller freely oscillating orbits, whereas a jump-up occurs from that branch to the tapping-mode branch during backward sweeps. This behavior is the reverse of that observed in the hysteric region located in the vicinity of primary resonance where the jump-up occurs during forward sweeps and the jump-down during backward sweeps. These distinctions raise questions about the nature of those large oscillations away from resonance and the nature of the pull-in instability they trigger.

Quasi-Static Pull-in - Actuator #1. We compare this actuator's response under the same voltage waveform $V_{ac} = V_{dc} = 6.863$ V at two excitation frequencies $f = 10$ and 25 kHz. The velocity and displacement time-histories of the actuator tip are shown in Fig. 3(a,d). Planar projections of the corresponding phase-portraits are shown in Fig. 3(b,e) while a 3-D phase-portrait of the orbit at $f = 10$ kHz is shown in Fig. 3(c).

The response at $f = 10$ kHz displays the large oscillations mentioned above unlike the case at $f = 25$ kHz. The time-history of the displacement and velocity in Fig. 3(a) shows that as the voltage varies, the actuator tip is pulled-in periodically coming into contact with the substrate where the displacement is held constant for a period of time before pulling off and settling down close to the equilibrium position. The phase-portrait at $f = 10$ kHz reveals a Shilnikov orbit homoclinic to a saddle-focus, Fig. 3(c). The orbit is composed of a flow along the unstable manifold away from the saddle, captured by impact and rebound on the substrate, reinserted into the vicinity of the stable focus, where it settles down, before the excitation waveform displaces it once again towards the saddle²⁷. The process of pulling-in corresponds to the flow along the unstable manifold of the saddle while settling down around the stable equilibrium corresponds to the stable focus. Flow again towards the saddle occurs due to the cyclic rise in instantaneous voltage $V(t)$ as it approaches the static pull-in voltage $V_{ps} \approx 15.60$ V.

We note that concepts of static pull-in do not apply to this process. The RMS of the voltage waveform $V_{\text{RMS}} = 11.716$ V is significantly less than V_{ps} . Further, the response is periodic and free of transients while static pull-in is fundamentally a transient phenomenon. Beyond the bifurcation point at the peak of the tapping branch, the actuator jumps down to a stable forced response orbits as shown in Fig. 3(d,e) at $f = 25$ kHz with the orbit size along the displacement-axis shrinking from $2 \mu\text{m}$ at $f = 10$ kHz to $1.120 \mu\text{m}$ at $f = 25$ kHz.

Increasing the excitation voltage to $V_{ac} = V_{dc} = 7.725$ V reduces the frequency domain where Shilnikov orbits and large oscillations occur. Experimentally measured velocity and displacement time-histories and the corresponding phase portraits at excitation frequencies of $f = 6.40$, and 7.0 kHz are shown in Fig. 4 for time spans of 2.560, and 7 excitation periods, respectively.

While a similar behavior to that described above is observed here, the bifurcation point is lactated at the peak of the tapping branch $f = 6.49$ kHz. Similarly, the orbit size shrinks from a displacement of $2.5 \mu\text{m}$ at $f = 6.4$ kHz to $1.1 \mu\text{m}$ at $f = 7.0$ kHz. While oscillations typical of a stable focus can be observed in Fig. 4(e), they result from the actuator rebounding from its maximum deflection point as can be seen in the corresponding time-history, Fig. 4(d). The complexity of the orbit is merely a reflection of the interaction between the slow time-scale of forcing and the fast time-scale of the actuator's fundamental mode. No tapping or flow along the unstable manifold is observed in Fig. 4(d,e). The fast-slow dynamics in this region result in two distinct trains of peaks in frequency-domain, corresponding to the forcing frequency and to the fundamental natural frequency of the actuator. For this waveform, we observed fast-slow dynamic interactions in the region extending from the bifurcation point ($f = 6.49$ kHz) until approaching the superharmonic resonance of order-three ($f \approx 25$ kHz).

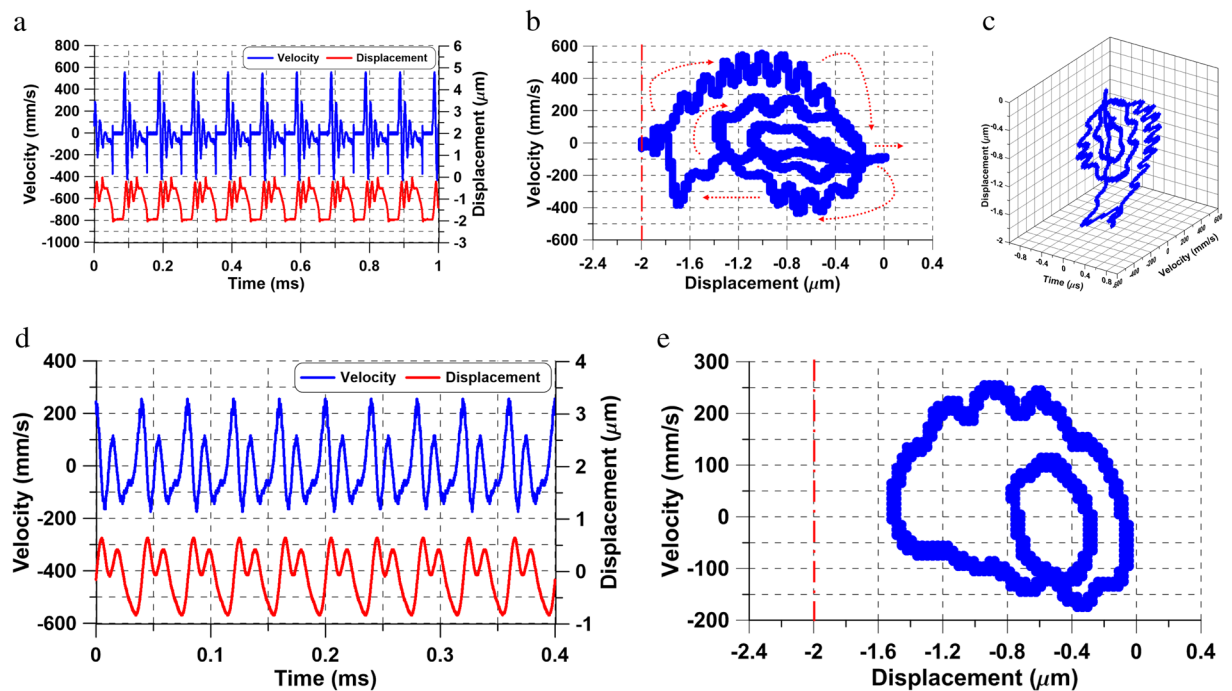


Figure 3. The time-histories of actuator # 1 tip velocity (blue line) and displacement (red line) under the voltage waveform $V_{ac} = V_{dc} = 6.863$ V and a signal frequency of (a) $f = 10$ kHz and (d) $f = 25$ kHz. The corresponding 2-D phase portraits (b) at $f = 10$ kHz, and (e) $f = 25$ kHz. (c) The 3-D phase portrait reconstructed from the time-histories shown in (a).

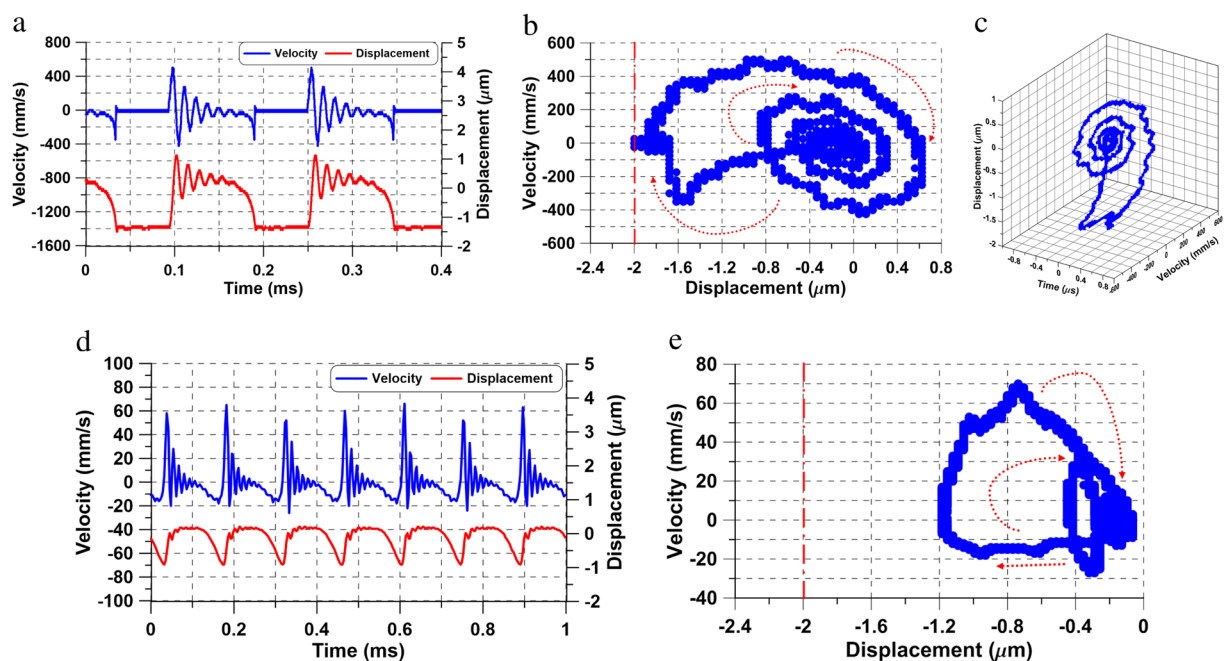


Figure 4. The time-histories of actuator # 1 tip velocity (blue line) and displacement (red line) under the voltage waveform $V_{ac} = V_{dc} = 7.725$ V and a signal frequency of (a) $f = 6.40$ kHz and (d) $f = 7.0$ kHz. The corresponding 2-D phase portraits (b) at $f = 6.40$ kHz, and (e) $f = 7.0$ kHz. (c) The 3-D phase portrait reconstructed from the time-histories shown in (a).

Quasi-Static Pull-in - Actuator # 2. We compare this actuator's response under the voltage waveform $V_{ac} = V_{dc} = 6.750$ V at two excitation frequencies $f = 5$ and 10 kHz. The tip displacement and velocity time-histories and the corresponding phase-portraits are shown in Fig. 5. Shilnikov orbits homoclinic to a saddle

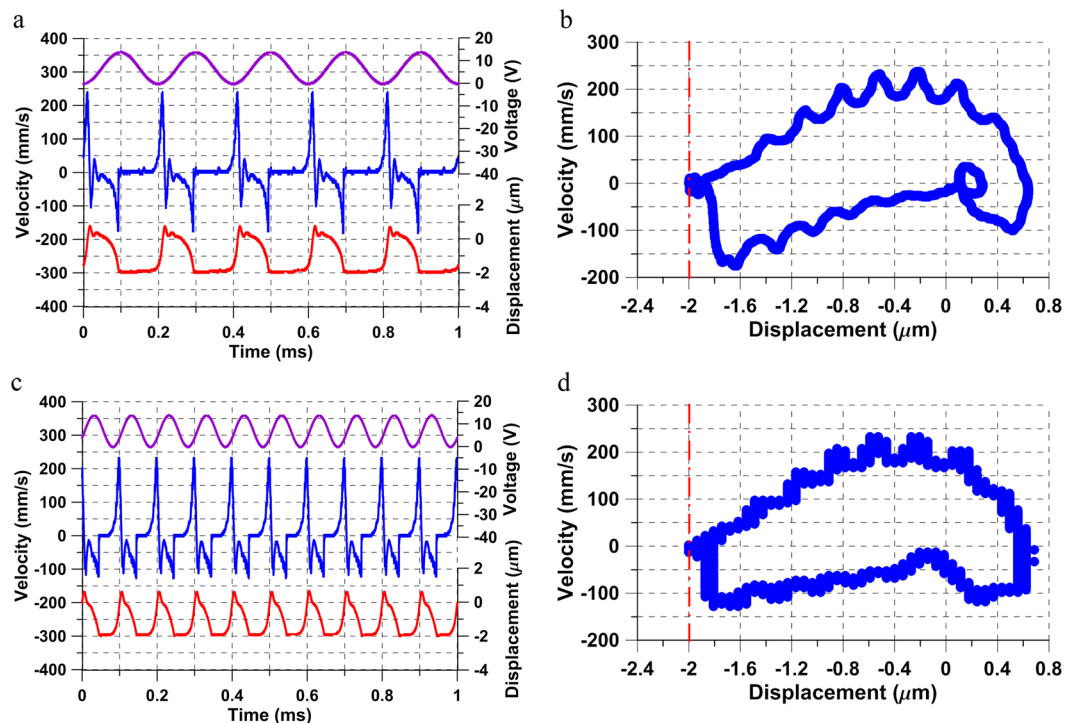


Figure 5. The voltage waveform (magenta line) and time-histories of actuator # 2 tip velocity (blue line), and displacement (red line) under the voltage waveform $V_{ac} = V_{dc} = 6.750$ V and a signal frequency of (a) $f = 5.0$ kHz and (c) $f = 10$ kHz. The corresponding phase portraits at (b) $f = 5.0$ kHz and (d) $f = 10$ kHz.

focus were observed in both cases, Fig. 5(b,d). The stable focus oscillations occur over a shorter settling time because of the actuator's lower quality factor ($Q = 2.1$). Comparison of the voltage waveform to the displacement time-history, Fig. 5(a,c), shows that the actuator pulls-in as the instantaneous voltage crosses the value of static pull-in, ($V_{ps} \approx 12.15$ V). Subsequently, it maintains contact with the substrate until the instantaneous voltage drops below pull-off voltage.

The voltage waveform was increased to $V_{ac} = V_{dc} = 7.650$ V for the excitation frequency $f = 10$ kHz, Fig. 6, which resulted in Shilnikov chaos. The velocity and displacement time-histories of the chaotic tapping oscillations are shown in Fig. 6(a). The corresponding phase portrait, Fig. 6(b), shows a banded chaotic attractor. This chaotic attractor was found to extend over the frequency [10–18] kHz during a forward sweep.

To estimate the Largest Lyapunov Exponent (LLE), a nonlinear autoregressive moving average model (NARMAX) was first identified from the velocity time-history. A comparison of the model predicted (red lines) and measured (blue lines) velocity over twenty excitation periods is shown in Fig. 6(c) to confirm the model validity. The logarithmic lower bound error (LBE) between two equivalent extensions of the model²⁸ is shown in Fig. 6(d). The LLE was estimated from the slope of the line ($LBE = 0.004t - 43.67$) approximating the initial growth of LBE as 2×10^{-6} bit/s. The positive LLE establishes that the underlying attractor is chaotic.

Discussion and Conclusions

We identified a new type of pull-in instabilities in electrostatic actuators dubbed quasi-static pull-in. It occurs under periodic excitation, as opposed to the case for static pull-in, in a non-resonant frequency range much lower than the fundamental natural frequency $ff_1 < 1$, as opposed to the case for dynamic pull-in. The instability was replicated and verified in two independent MEMS electrostatic actuators oscillating in ambient air.

The transition from static pull-in to quasi-static pull-in occurred in both cases at 5 kHz and corresponded to the beam tip contact with the substrate transitioning from area-contact to line-contact. In all cases, the peak instantaneous voltage $V(t)_{Max}$ was close to but below static pull-in voltage, thereby allowing transients to cause pull-in. To analyze this transition, we write the force balance driving the actuator's upward acceleration \ddot{x} as:

$$m\ddot{x} = F_{rs} + F_{rb} - F_e - F_s \quad (1)$$

where m is the effective mass, F_{rs} restoring force, F_{rb} rebound force recovered from momentum before impact, F_e electrostatic force, and F_s net surface stiction force. We further note that F_{rs} and F_s are functions of the contact condition rather than the voltage waveform. Both forces are larger when area-contact develops than they are when line-contact prevails²⁹. On the other hand, F_{rb} increases with the excitation frequency as the velocity of the actuator increases and F_e is proportional to the square of the waveform $V^2(t)$ and larger in area-contact than line-contact.

At lower frequencies, the rebound force F_{rb} is low and the electrostatic force F_e drops slowly towards zero, thereby allowing the actuator to evolve into area contact and resulting in a larger surface contact force F_s . The

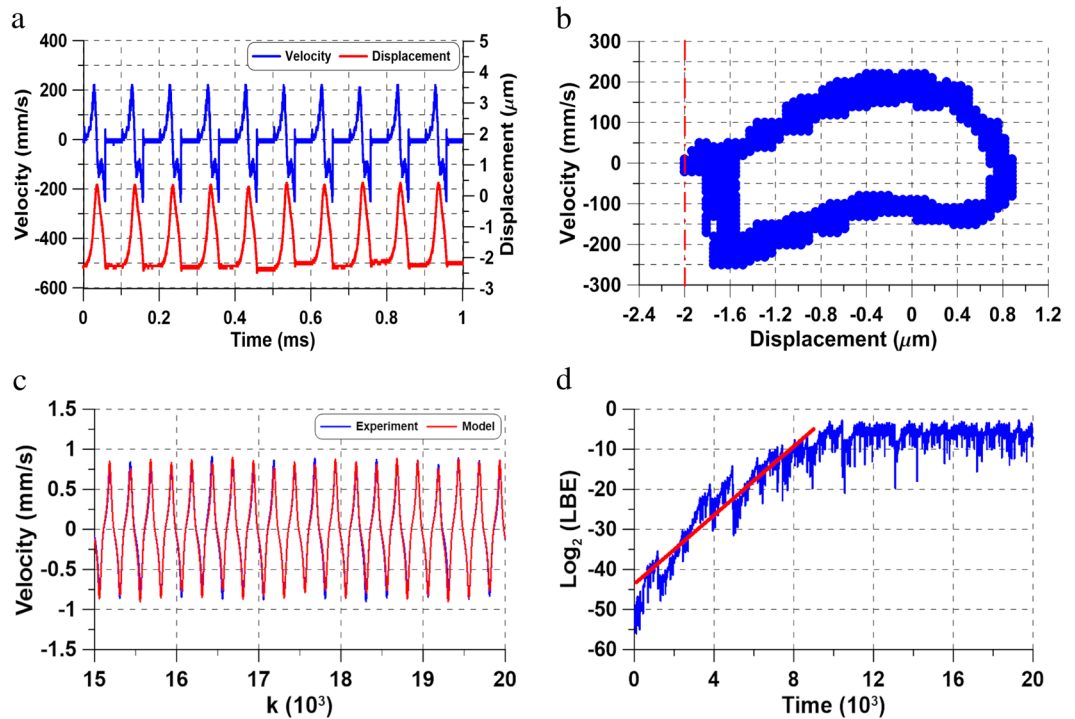


Figure 6. (a) The time-histories of actuator # 2 tip velocity (blue line) and displacement (red line) under the voltage waveform $V_{ac} = V_{dc} = 7.650$ V and an excitation frequency of $f = 10$ kHz, (b) the corresponding phase portraits, (c) comparison of measured (blue line) and simulated (red line) velocity time-histories and (d) computation of the largest positive Lyapunov exponent (LLE).

force balance in this case $F_{rs} + F_{rb} - F_e - F_s < 0$ prevents pull-off, and therefore motion. Beyond 5 kHz, the rebound force F_{rb} is larger and the electrostatic force drops faster towards zero preventing area-contact and limiting the surface contact force to smaller values. The resulting force balance $F_{rs} + F_{rb} - F_e - F_s > 0$ favors pull-off.

Therein lies a fundamental difference between static pull-in and quasi-static pull-in: transient (non-repeatable) dynamics underlie the first whereas orbital dynamics underlie the second. Quasi-static pull-in requires oscillatory forcing at a frequency high enough to initiate pull-off. A necessary condition for that are low surface stiction forces.

Quasi-static pull-in is driven by a fast-slow dynamic interaction between slowly varying electrostatic excitation and fast response of the actuator's fundamental mode. It is triggered by excitation waveforms where the instantaneous voltage approaches or exceeds static pull-in voltage V_{ps} resulting in the appearance of Shilnikov orbits homoclinic to a saddle-focus. Those orbits are characterized by large tapping-mode oscillations where the actuator periodically goes to pull-in through the saddle (unstable equilibrium), pulls-off and settles around the stable equilibrium, before drifting again towards the saddle to repeat the cycle. The actuator's quality factor determines the order of the homoclinic orbit (settling time around the static equilibrium)³⁰. The homoclinic orbit observed in actuator # 1 ($Q = 5.4$) was of order 5, whereas those observed in actuator # 2 ($Q = 2.1$) were of order 1.

In forward frequency-sweeps, Shilnikov homoclinic orbits appear beyond static pull-in, persist along a branch of tapping-mode oscillations, and evolve into Shilnikov chaos. As the excitation frequency increases, the impact speed increases, thereby increasing impact-induced damping. Elevated damping reduces the real part of the saddle's leading pair of complex eigenvalues ρ , therefore violating the condition for Shilnikov homoclinic orbits: $\delta = \rho/\lambda < 1$. At that point, the branch terminates with the response falling down to a coexisting branch of stable freely oscillating periodic orbits. A Shilnikov bifurcation demarcates the end of the free oscillations branch where the response jumps-up to the tapping-mode branch during backward frequency-sweeps. A hysteretic region exists between these jumps, bounded at the lower-end by the jump-down in forward-sweeps and at the upper end by the jump-up in backward-sweeps. This is another characteristic of quasi-static pull-in in contrast to the hysteretic region associated with dynamic pull-in where the jump-down occurs in backward-sweeps and the jump-up in forward-sweeps.

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Author contributions

M.S. Al-Ghamdi carried out the experiments, analyzed the results, and wrote the manuscript. M. Khater designed the experiments, analyzed the results and wrote the manuscript. E.M. Abdel-Rahman designed the experiments, analyzed the results and revised the manuscript. E.G. Nepomuceno modeled the largest Lyapunov exponent and revised the manuscript.

Competing interests

The authors declare no competing interests.

Additional information

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